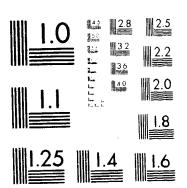
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SCHEDULING OF VARIABLE INLET GUIDE VANES,
GAS GENERATOR SPEED, AND POWER TRAIN
NOZZLE ANGLE ON THE PEPTORMANCE OF AN
AUTOMOTIVE GAS TURBINE ENGINE

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POWER TURBINE NOZZLE ANGLE

ON THE PERFORMANCE OF AN

AUTOMOTIVE GAS TURBINE ENGINE

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SUMMARY

The engine power augmentation, specific fuel consumption, and emissions obtained with water injection downstream of the compressor were investigated over a range of variable inlet guide vane settings, gas generator speeds, and power turbine nozzle settings on the Chrysler/ERDA Baseline Automotive Gas Turbine Engine. The experimental results were used to estimate the potential fuel economy of a vehicle powered by a Chrysler/DOE Upgraded engine augmented with variable geometry and water injection.

Results indicated the amount of power augmentation achieved depends upon the method of utilizing the variable geometry. Power increases of over 20% and SFC reductions of over 5% were obtained with water injection and variable geometry utilization if gas generator speed was varied to keep turbine inlet temperature constant. When gas generator speed was held at 95% of design, the power augmentation was about 16% and SFC reduction was about 4%. The hydrocarbons and carbon monoxide emissions were not influenced by the water injection, but the oxides of nitrogen in the emissions were reduced by as much as 35%.

The results indicate at least a 5% improvement in fuel economy might be achieved over a composite (55% city/45% highway) Driving Cycle using a 75KW (100 HP) engine which could be augmented to 89 KW (120 HP) relative to an 89 KW (120 HP) unaugmented engine. The vehicle powered by the smaller, augmented engine would have the same engine.

INTRODUCTION

The power from an automotive gas turbine engine can be augmented through the use of variable geometry and injection of water downstream of the compressor (and upstream of the regenerator). An experimental investigation was conducted to determine the amount of power augmentation that can be obtained. The Chrysler/ERDA Baseline regenerative automotive gas turbine engine was used for these tests.

A characteristic of the gas turbine engine is to run efficiently over only a narrow speed range at or close to its maximum speed and maximum turbine inlet temperature. This is a severe hand cap to its application as an automotive powerplant which must be efficient over a broad range coinciding with the prevalent driving speeds of the automobile. To reduce this handicap, the efficient operating speed range has to be either broadened or shifted downward toward the midspeed range. The approach is to design the engine for less power, this narrow efficient operating range downward. However, this severely limits the driveability (acceleration) of the automobile. This performance limitation can be reduced by augmenting the power of the less powerful engine with:

- (1) variable inlet quide vanes;
- (2) with both water injection and variable inlet guide vanes;or
- (4) with both water injection and variable power turbine nozzle angle.

Tests on aircraft engines in the 1950s and, more recently, tests on the Chrysler/ERDA Baseline gas turbine engine by Chrysler, have demonstrated that a gas turbine engine output power can be efficiently augmented by as much as 10% by injecting water into the engine inlet. This is reported in Reference 1. Unfortunately, the amount of water necessary to provide this amount of power augmentation has led to the erosion of the compressor impeller blades. This was noticed in both the aircraft and automotive tests. im aircraft engine manufacturers overcame this problem by injecting the water interstage or downstream of the compressor. Reference 3 discusses the injection of water interstage on an afterburning engine to solve the problem of compresor casing thermal contraction caused by injecting water upstream of the It was suggested that to solve the problem of compressor impeller erosion the water be injected downstream of the compressor even though work reported in Reference 4 indicated that water injection rates required for water injection downstream of the compresor are about twice as great as for compressor inlet water injection.

The background, function, purpose, and test results of the use of variable inlet guide vanes to augment the maximum power of the Cnrysler/ERDA Baseline Gas Turbine engine are presented in Reference 2. Some of the VIGV tests reported in Reference 2 were also conducted in this investigation to establish a base for comparison to the water injection results.

This investigation was conducted to complement the Chrysler test program reported in references 1 and 2. An engine similar to the one described in reference 1 was modified by installing water spray nozzles downstream of the compressor in the diffuser section. This location is upstream of the regenerator. No other modifications were made to the engine except for the addition of instrumentation needed to determine output power, specific fuel consumption, emissions, and various temperatures and pressures in the engine.

All data were taken at steady-state conditions. The engine was allowed to stabilize for at least ten minutes at each point before data was recorded. The basic gas generator speed investigated was 95% of design. The influence of the variable inlet guide vanes was investigated over the range of -20° to +30° without water injection and from 0 degrees to -30 degrees with water injection. Water was injected at rates that resulted in water/air mass ratios of .0076, .023, and .039. Power turbine nozzle angle could not be measured; however, the same power turbine nozzle setting was insured by always

operating the engine at the same dry reference point prior to any water injection. The output shaft speed was held constant (3473 ± 9) RPM) for all data points. Table I lists actual selected engine operating parameters for all the test points.

One of the power augmentation procedures examined in this investigation would require gas generator speeds higher than design. This would tend to reduce the life of the engine as designed but it was investigated to determine its effectiveness. Designers of new engines might be able to utilize it without any significant compromise in durability or rotor inertia.

The author acknowledges the assistance of John L. Klann of the Systems Analysis and Assessment Office at Lewis Research Center for the theoretical prediction of SFC vs HP shown in Figure 15 and for theoretical projections of fuel economy.

SYMBOLS

CO Carbon Monoxide

CO₂ Carbon Dioxide

f/A Fuel/Air Ratio

HC Hydrocarbon Emissions

HP Power Output

ND Dynamometer Speed

NGG Gas Generator Speed

NO_x Oxides of Nitrogen

P Pressure

PPM Parts Per Million

SFC Specific Fuel Consumption

T Temperature

VIGV Variable Inlet Guide Vanes

WAC Air Mass Flow Rate

WFC Fuel Flow Rate

WWC Water Flow Rate

Station Notation is shown in Figure 4.

ENGINE DESCRIPTION

The engine is a low pressure ratio, regenerative, free-power turbine design. It incorporates a single-stage centrifugal compressor with variable inlet guide vanes, a single can-type combustor, an axial turbine stage to drive the compressor, and a free power-turbine with variable nozzle blading. A drawing of the basic engine arrangement is shown in Figure 1. Figure 2 shows the engine installed in the test stand. Additional information on the engine is contained in

Distilled water was used in the tests. There were two supply lines - one for each side of the engine. Each contained a turbine type flow meter, and each supplied water to a furnace-type simplex burner nozzle. A photograph of the engine showing the location of one of the nozzles is shown in Figure 3.

The station notation convention used for the engine is shown in Figure 4. Both total temperature and pressure were measured at stations 1, 2, 6, and 8, and the exhaust duct. In addition, station 4 pressure and station 5 temperature were measured. Also measured were gas generator speed, dynamometer speed, air flow, fuel flow, water flow, and variable inlet guide vane angle. Mechanical difficulties made it impractical to measure the power turbine nozzle angle.

The inlet air temperature was held constant at 30°C (85°F) for all data points by the facility air system. The power turbine speed was controlled by the dynamometer and was held constant at 3473 + 9 independently controlled, and the water flow was also independently controlled.

In runs in which gas generator speed was set, the engine fuel control automatically adjusted fuel flow to maintain that speed. This important to keep in mind when rationalizing events during water injection when power is increased by varying inlet guide vanes and/or the power turbine nozzles.

During most of the runs it was necessary to hold the power turbine nozzle angle at a fixed value. To overcome the previously mentioned problem of measuring this angle, the engine was operated at the same dry reference point prior to any series of runs. At this dry reference point the power turbine nozzle position was set by varying the nozzle until T8 was 705°C (1300°F). With engine inlet temperature, burner exit temperature, gas generator speed, and dynamometer speed set at the constant dry reference point conditions; T8 control was a very accurate method of assuring the same position of the power turbine nozzles.

TEST PROCEDURE

The constant dry reference point nominal values were: VIGV angle = 0; NGG = 95% design (design equals 44,610 RPM); and Tg = 705°C (1300°F); burner exit temperature 942°C (1727°F). The 95 percent of design gas generator speed point was selected as the dry reference data point, to allow speed and temperature excursions beyond the dry reference point with little risk of engine damage. Selected actual engine operating parameters at the dry reference point are shown in Table 1, items 32, 39, 44, 54, 62, 70, and 90.

Starting at the dry reference point, the following tests were conducted to determine the effect on power, SFC, and emissions:

- Varying the inlet guide vanes with no water injection.
- 2. Water injection.
- 3. Varying the inlet guide vanes with water injection.
- 4. Varying both the inlet guide vanes and the power turbine nozzle with water injection.
 - 5. Varying the gas generator speed with no water injection.
- 6. Varying both the inlet guide vanes and the gas generator speed with water injection.

Varying the Inlet Guide Vanes (no water injection). The inlet guide vanes were varied from -20 to +30 degrees with data recorded every five degrees. During the first series of runs burner exit temperature was not controlled and from -20 to 0 degrees the burner exit temperature exceeded the reference value. Additional tests were then made from -30 to 0 degrees VIGV angle settings and the burner exit temperature was maintained constant at the dry reference value by also varying the power turbine nozzle angle. (A positive sign of the inlet guide vane setting indicates the incoming air is directed into the direction of rotation of the compressor rotor. A negative sign indicates the air is directed opposite to the direction of rotation of the compressor rotor).

<u>Water Injection</u>. After data were recorded at the dry reference point water was injected at water/air flow ratios of .0076, .023, and .039. After allowing enough time for burner exit temperature to stabilize, the water injection data point was recorded. No other changes were made to the operating condition of the engine.

Varying the Inlet Guide Vanes with Water Injection. After the first water injection data point was recorded at 0° inlet guide vane setting, the inlet guide vanes were varied 5 degrees at a time, and data points were taken until the burner exit temperature matched the dry reference temperature of 942°C (1727°F). This procedure was repeated for the other two water/air flow ratios.

Varying Both Inlet Guide Vanes and Power Turbine Nozzle With Water Injection. The inlet guide vanes were varied and data were recorded every 5 degrees, while water was injected. Burner exit temperature was held equal to its dry reference value by varying the power turbine nozzle angle. This procedure was repeated for the other two water/air flow ratios.

Varying Gas Generator Speed ($N_{\rm GG}$) No Water Injection. Starting at the dry reference point, the gas generator speed was increased and data was recorded at points corresponding to 96, 97, and 97.6 percent of gas generator speed. No other changes were made to the engine.

Varying Both Inlet Guide Vanes and Gas Generator Speed with Water Injection. The inlet guide vanes were varied and data were recorded every 5 degrees, while water was injected. The burner exit temperature was held equal to its dry reference value by varying the gas generator speed. The power turbine nozzle angle was held constant at the dry reference value. This procedure was repeated for the other two water/air flow ratios.

RESULTS AND DISCUSSION

Presented in the following sections are the results of the separate and combined effects that VIGV angle, variable power turbine nozzle angle, speed, and water injection had on power output, specific fuel consumption, and emissions. Because of difficulties in measuring the power turbine vane angle, no attempt was made to determine the SFC and power changes that could be obtained by varying only the power turbine nozzle angle. All data points are shown in Table 1.

Power Output and Specific Fuel Consumption

Varying the Inlet Guide Vanes With No Water Injection. The result that varying the inlet guide vane angle with no water injection had on power output is shown in Figure 5a. If the VIGV angle is the only variable, and if no attempt is made to limit burner exit temperature, the power output continues to rise at almost a steady rate up to a VIGV angle of at least -20 degrees. At -20 degrees there is a 9% increase in power. If the burner exit temperature is limited to the dry reference point value of 942°C (1727°F) for all negative values by varying the power turbine nozzle, the power output will increase by about 2.5% then start to decrease at about -15 degrees VIGV angle.

The relationship of power output change to burner exit temperature change over the range of variable inlet guide vane angles investigated is shown in Figure 5b.

The specific fuel consumption for the above data points is shown in Figure 5C. Limiting the burner exit temperature results in higher SFC values.

The data points for Figure 5 are 32 to 51 in Table 1.

Water Injection. The injection of water only without allowing the VIGV angle, the power turbine nozzle angle, or the gas generator speed to vary from their dry reference value caused no significant change in power or SFC as shown in figure 6a and 6b. However, it did decrease the burner exit temperature significantly as shown in figure 6c. The data points for figure 6 are 55, 71, and 91.

Varying the Inlet Guide Vanes with Later Injection. For water injection at various water/air flow ratios and inlet guide vane settings, the changes in burner exit temperature from the dry reference value of 942°C (1727°F) and the changes in engine power are shown in Figure 7a. Power increases of up to 13% at the maximum water/air flow ratio (Point 96) were observed. Gas generator speed was maintained at 95% of design.

The effects of water injection alone are noted as points 55, 71, and 91 on figures 7a (these points are shown previously in figure 6). Data points with water injection and with the VIGV angle varied to maintain burner exit temperature equal to the dry reference point value of 942°C (1727°F) are noted as points 58, 75, and 96 on Figures 7a, and 8a. From Figures 7a and 8a, it is apparent that these power increases with water injection are almost the same as those obtained at equal VIGV angle settings without water injection. Further when the points from the highest water/air flow ratio curve from Figure 7a are placed in Figure 8b along with the dry points, it is observed that the power increases are the same for corresponding temperature changes for both the dry and the water injected points.

From Figures 8a and 8b it is concluded that the power increase must be due only to the inlet guide vane angle change. The water injection merely lowered the temperatures, thus making it possible to vary the inlet guide vane angle without exceeding the dry reference point temperature of 942°C (1727,F).

Varying the VIGV angle setting with various water/air flow ratios not only increased power as noted in Figure 8a but effected SFC as shown in Figures 7b, 7c and 7d. The overall trend is to lower SFC values with both increased water injection ratios and increasingly negative VIGV angles. The VIGV angle appears to be the more prominant effect as shown by Figure 7a, 7b, and 7c. When the water injection points 58, 75, and 95 are plotted with the dry VIGV angle points as shown in Figure 8c, it becomes apparent that the changed VIGV angle is responsible for the reduction in SFC (up to 1.5%). The water injection merely allowed the engine to be rematched at a more negative VIGV angle setting which improved both the power and SFC, as noted in Figures 8a and 8C, but without going to the higher burner exit temperatures indicated on Figure 5 b for the more negative values of VIGV without water.

The data points for Figure 7 are 54 to 58, 71 to 75, and 91 to 96. The data points for Figure 8 are 32 to 43, 54, 58, 75, and 91 to 96.

Varying Both the Inlet Guide Vanes and the Power Turbine Nozzle with Water Injection. Shown in Figure 9a are the power changes obtained without water injection and at the three different water/air flow ratios by varying the VIGV angle with the burner exit temperature held to the dry reference point value of 942°C (1727,F) by varying the power turbine nozzle angle. Power increases of up to 16% were observed.

For the above points, the specific fuel consumption as a function of VIGV angle is shown in Figure 9b, which shows a reduction of up to 4.5%.

Varying the Gas Generator Speed Without Water Injection and Without Any Engine Geometry Change. The results that varying the gas generator speed had on power output without water injection or engine geometry change are shown on Figure 10a, and the relationship of power output change to burner exit temperature increase is shown in Figure 10b.

The specific fuel consumption as a function of gas generator speed is shown in Figure 10c.

The data points for Figure 10 are 164 to 167.

Varying Both the Inlet Guide Vanes and the Gas Generator Speed with water Injection. Shown in Figure 11 are the power change results plotted as a function of VIGV angle for the three different water/air flow ratios investigated. The burner exit temperature was held at the dry reference point value of 942°C (1727°F) by varying the gas generator speed. Power increases up to 23% were observed. Because varying the VIGV angle did not increase the power, the remainder of this section concerns only the results at 0 VIGV angle.

Figure 12a shows that the same amount of power augmentation was obtained with and without water injection by varying gas generator speed. Water injection permitted burner exit temperature to be held constant. In Figure 12b, power change is plotted against the absolute value of the change in burner exit temperature IT51. This figure shows three sets of data:

- Water injection only (points 55, 71, 91) (12b only)
- Power augmentation by varying gas generator speed only.
- 3. Power augmentation by water injection and varying gas generator speed to hold the burner exit temperature constant (Point 66, 85, 101).

The results show that the power augmentation caused by varying only

gas generator speed is at the intersection of the other two sets of data. From figures 12a and b, it is apparent that the power increase results from increasing gas generator speed to rematch the engine to the dry reference burner exit temperature after that temperature was lowered by water injection.

The specific fuel consumption change is shown plotted on Figure 12c along with the dry points of Figure 10c. The results indicate that the rate of improvement in SFC is about the same as obtained by allowing the burner exit temperature and the gas generator speed to increase without water injection.

The power increase and the SFC improvement, therefore, appear to be due only to the gas generator speed increase necessary to rematch the engine to its dry reference burner exit temperature. The water injection by itself did not improve the power output or the SFC.

The limit on the amount of power augmentation possible with water injection was not reached during this investigation. Although it is certain there is a limit on the amount of water that can be effectively injected, no limit was found within the scope of this investigation. However, the slight increase in power between the two highest water flow rates indicates that a limit is being approached on the amount of power augmentation possible from varying the inlet quide vanes with water injection. This is due to the reduced influences of inlet guide vane angles on power augmentation past -20 degrees. Higher water injection rates were not investigated because there was concern about surging the compressor.

Emissions

The injection of water caused a significant reduction in NOx emissions, about a 35% for the .039 water/air flow ratio. There was also : further reduction in NO $_{\rm X}$ associated with the reduction in the burne. exit temperature of about .7% for each degree centrigrade. The percent reduction was calculated based on the dry reference point taken immediately preceding the water injection points, and is plotted against the reduction in burner exit temperature in Figure 13. Data in Figure 13 includes every water injection test point and are points 26 to 108 in Table 1.

A slight reduction in CO and HC was also observed. However, the reduction was small enough to be in the error band of the measuring equipment; therefore, no attempt was made to correlate the changes in these emissions with water injection rate, gas generator speed, or changes in engine geometry.

The emissions data are presented in Table 1.

Application to the Automobile

To determine the possibilities of the application of power augmentation by water injection to an automobile, the following example was assumed:

- 1. 1588 kg (3500 pound) automobile.
- 2. 89 kw (120 HP) engine.
- 3. 20% augmentation of the max. engine power is possible by some combination of water injection, variable geometry, and gas generator overspeed which will not impair engine life.

The amount of power needed to propel the automobile over either the highway or city driving cycles is about half that needed to provide a desired 0 to 97 km/h (60 MPH) acceleration of the automobile. For example, the 1588 kg (3500 pound) automobile requires 50 kw (67.568 HP) maximum engine power for the urban driving cycle, but requires 89 kw (150 HP) to accelerate from 0 to 97 km/h (60 MPH) in 13 seconds. Figure 14 shows a typical relationship between power per unit weight of the automobile and the time to accelerate from 0-97 km/h (60 MPH). For this example, it has been assumed that 89 kw (120 HP) is needed to provide acceptable acceleration characteristics for the 1588 kg (3500 pound) automobile. If this 89 kw (120 HP) can be provided by an engine nominally designed for 75 kw (100 HP) and augmented by some combination of variable geometry, gas generator overspeed, and water injection to 89 kw (120 HP) then fuel savings can be realized by using the smaller engine which will operate closer to its optimum operating conditions than the larger engine.

In addition, it is expected that vehicle weight will be reduced due to using an augmented 75 kw (100 HP) engine in place of a 89 kw (120 HP) engine. Advanced engines will weigh approximately 1.5 kilograms per kilowatt (2.5 pounds per horsepower) and this is assumed to be the weight that will be saved by using a less powerful engine. Applying this assumption of 1.5 kilograms per kilowatt (2.5 pounds per horsepower) to the example results in a 22 kg (50 pounds) savings due to engine size reduction. Some or all of these savings will be offset by the weight of the water, water tank, and injection system required. Assuming a 2.3 kg (5 pound) water system approximately 20 liters (5.4 gallons) of water could be carried without exceeding the 22 kg (50 pounds) that was saved.

The 55% City/45% Highway FDC composite fuel economy for a 1588 kg (3500 pound) vehicle was generated assuming the SFC vs. HP characteristics for the 75 kW (100 HP) and 89 kW (120 HP) engines shown in Figure 15. The results were 16.7 KM per liter (34.1 MPG) and 15.6 km per liter (31.9 mpq) respectively; a gain of 6 percent for the smaller engine. When the 75 km (100 HP) engine is power augmented by water injection, it takes on the acceleration and driveability characteristics of a 89 kw (120 HP) engine with the fuel economy of a 75 kw (100 HP) engine. Therefore, downsizing the engine to 75 kw (100

HP) results for this example in a fuel economy gain of about 6 percent while meeting the power required for acceleration through augmentation. Since power augmentation was not necessary to execute the 55% Citv/45% Highway FDC composite driving cycle with the smaller engine, no estimate of water requirements could be made. However, the amount of water required by less conservative driving habits should not be excessive due to the short duration of augmentation demand.

For the purpose of illustration, 89 kw (120 HP) was selected as the minimum power desired for the 1588 kg (3500 pound) automobile, but the example and the trend toward improved fuel economy may be valid for other engine powers and automobile weights.

CONCLUSION

The power output of an automotive gas turbine engine can be augmented through the separate or combined use of variable geometry gas generator overspeed and injection of water downstream of the compressor (upstream of the regenerator). An experimental investigation using the Chrysler/ERDA Baseline Automotive Gas Turbine Engine was conducted to determine the amount of power augmentation that can be obtained. The base case for all tests was at 95% of design gas generator speed.

Varying only the VIGV angle to -20 degrees produced an increase in power of 9.5% and a decrease in SFC of 1.5%. At the same time however, the burner exit temperature was required to increase.

Varying only the gas generator speed from 95% to 97.6% of design produced an increase in power of 15%, and a decrease in SFC of 4%. At the same time however, the burner exit temperature was required to increase.

Water injection downstream of the compressor resulted in a reduction in burner exit temperature making possible a rematching of the engine back to the original (dry) burner exit temperature. The rematched conditions resulted in an increase in output power and a reduction in SFC without exceeding the base case burner exit temperature. The engine was rematched by the following:

- Varying the inlet guide vanes;
- 2. Varying the power turbine nozzle;
- 3. Increasing the gas generator speed above design;
- Various combinations of the above.

At a water/air flow ratio of .039 a gas generator speed of 95% of design and a base case burner exit temperature of 942°C (1727°F) the following results were obtained:

1. Varying the inlet guide vanes produced a power augmentation of 13.5% and an SFC reduction of 1.5%.

- 2. Varying the power turbine nozzle produced a power augmentation of 10%, and an SFC reduction of 3.2%.
- 3. Varying both VIGV and power turbine nozzle produced a power augmentation of 15.8% and an SFC reduction of 4.5%.

At water/air flow rate of .039 increasing gas generator speed to maintain base case burner exit temperature produced a power augmentation of 23%, and an SFC reduction of 5.2%. Varying the inlet guide vanes provided no additional increase in power.

The above power augmentation was achieved with a 35% reduction in $^{\rm NO}{}_{\rm X}$ and without any increase in other pollutants.

When applied to a typical automobile a 6% improvement in fuel economy is predicted over a 55% City/45% Highway FDC composite driving cycle using a 75 kW (100 HP) engine which could be augmented to 89 kW (120 HP) relative to an 89 kW (120 HP) unaugmented engine. The vehicle powered by the smaller, augmented engine would have the same 0-60 mph acceleration characteristics as the vehicle with the larger engine.

REFERENCES

- Angell, Peter R.; and Golec, Thomas: Upgrading Automotive Gas Turbine Technology; An Experimental Evaluation of Improved Concepts, SAE Paper 760280, Feb. 1976.
- 2. Pampreen, R. C.: The Use of Variable Inlet Guide Vanes for Automotive Gas Turbine Engine Augmentation and Load Control. SAE Paper 760285, Feb. 1979.
- 3. Useller, James W.; Huntley, S. C.; and Fenn, David B.: Combined Compressor Coolant Injection and Afterburning for Turbojet Thrust Augmentation. NACA RM E54G08, 1954.
- 4. Lundin, Bruce T.: Analysis of Turbojet Thrust Augmentation Cycles. Institute of the Aeronautical Sciences Preprint #223, Mar. 1979.

RUN	N _{GG}	VIGV (°)	WNC Kg/HR	WAC Kg/Sec	NPC Kg/Sec	f/a	Power Output KW	SFC	2.42	
32	95.00	+0.37). >	2.963	0.00388	3,03°26	90.223	Eg/Kw-Hr	P_2/P_1	NDC (RPM)
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5.8	95.01	-11.96	26.3	0.975	0.00025	0.00944	94.123 94.772	0.352	3. 746	3469.7
5) 60	44. 32	-9.98	26.3	0.970	2.00918	3.93445	94.051	0.351 0.343	1. 334 3. 735	3468.2
e 1	94.91 95.12	-5:83		7.961	0-00104	0.00945	93.397	G. 150	75	3471.3 3469. 3
6.7	94.96		მო•5 ი•გ	3.956	3.00497	0.03934	92.967	0.347	3. 73.2	3476.4
6.1	95.02	-12.03	26.4	0.961 0.975	0.00435 0.00438	0.00920	89.449	0.356	3. 709	3471.4
64	95.06	-9.95	26.5	0.172	0.00913	0.10951 0.00948	95.527 95.515	0.350	3-603	3469.8
65	45.78	-4.90	26.5	0.972	7.00919	0.10945	95.368	0. 143 0. 147	1.787 3.791	3474.9
6 A 6 B	95.88 94.34	-0.18	36.6	0.972	7.00720	0.00945	94.819	0.343	784	3470.2
70	95.04	-0.66 -0.62	3.0	0.961	9-00486	0.00920	89.236	0.354	3.717	3470.6 3 476. 6
71	95.04	-0.69	41.1	0.962 0.955	0-00369 0-00369	0.00920	90. 30 2	0.355	3.729	3481.0
72	95.33	-5.05	80.7	136.0	0-00914	0.00930 0.00951	90.601 92.739	0.353	3. 736	3478.1
71 74	95.36	-10.02	97.7	0.949	0.00927	0.00957	95.095	0.355 0.351	3 <u>.</u> 759 3. 800	3473.5
75	95.00 94.95	-15.12	H-) - C	0.973	0.00947	0.00971	97.137	0.351	3.835	3475.3
76	95.37	-19.98 -15.02	40.5 40.5	0.974	0.00967	0-00493	99.525	0.350	3.857	3472.5 3473.8
77	94.94	-10.02	83.3	0.97C 3.959	0.00961 0.00951	0.00989	100.196	0+345	3.842	3477.7
74	95.10	-5.01	87.6	0.950	0.00935	0.00991 0.00982	9 9. 286 9 7. 976	0.345	3.404	3474.4
79 H0	95.07	+0.24	80.7	0.946	9.00924	0.00975	95.973	0.343 C.347	3. 771 3. 742	3478.2
H1	9 5.02 95.05	+0.26	H). f	0.940	0.00925	0.00982	97.147	0.343	3.742	3476.5
a p	45.45	-19.97 -14.96	भते.्4 न7.2	0.976 1.987	0.00955	0.00976	99.454	0.346	3.862	3476.9 3479.4
83	96.50		60.0	J. 995	0.00977 0.00983	0.00959 0.09949	101.905	C.315	1.434	3475.5
2.4	a6* aa		80.3	3.991	0.00935	2.02.092	103.379 102.407	ۥ 443 ۥ41	1. 7. 1	3479.3
₽5 96	97.68		°).1	1,492	3.39937	1, 10004	133.321	C. 34	4.416 3.500	3475.9 3477.1
37	95.0d 95.0n		43.1	0.956	0.00111	1.02951	279.474	6.367	1.723	3482.0
90	<u> </u>	+0.02 +1.05	3.0 (.0	0.963 0.960	1.00993	0.00036	44.747	0.36.2	3_711	3476.6
9.1	94.93		15.1	7.948	0.00491 0.00867	0.1/973 0.19033	90 <u>, 2</u> 89 90, 34 7	0.355	3-200	3471.7
9.	94.61		36.5	J. 95,7	0.00914	วั.อังษ์เล	91.853	0.35a 0.35a	1, 7.14 1, 734	3466.8
93 94	44.47		35.5	0.960	0.00033	1.03969	25.277	0.352	1.740	3466.5
45	94.94		35. 5	0.968	0.00155	0.90482	97,930	C. 151	1. 043	3466.8 3470.9
91	45.00		35.1 35.0	0.466 0.466	7.00974 7.00988	0.01001	100.705	0. 146	3.370	3466.1
47	95.95		34.3	3.486	0.01307	0.01021	162.191	0.344	3. 184	
93	96.77	-15.02	*q. 7	0.996	0.01021	0.01022	100.471 108.766	C.340 C.334	1.951	3465.9 3466.0
9.9 100	97.26 97.61		14. 3	0.494	0.01025	0.11027	107, 729	0.143	3.943 3.944	3468.7 3469.7
101	48.44		14. /	0.447	0.01030	0.01031	108.479	0.142	1.995	3468.4
10.2	44.46		34.7 36.4	1.001 0.932	0.01034 0.00954	0.01027	110.832	0.136	4-023	3477.4
103	94.51	-5.07	36.3	0.938	0.00957	0.01021 0.01019	98.975 100.759	0-347	3-743	3468.1
104	95 . 15	-10.06 1	36.1	0.951	0.00969	0.01017	103. 106	0_342 0_33H	1.765	3465.2
105 106	95.17 94.75		3n. 1	0.462	0.00982	0.01016	104.264	0.339	3.855	3468.4 3467.3
107	94.75 95.00		16. J	0.9€2	0.00990	0.01029	102.981	C-340	3. 253	3463.7
164	94.97	-0.13	30.7	0.965 0.959	5PP00.0	0.01025	103.802	0-341	3. 472	3469.2
16.5	90.03	-0.12	3.3	0.975	0.00908 0.00950	0.00947 0.00975	40.025 95.582	0. 163	3705	3476.2
16 o	97.06	-0.13	0.3	0.991	0.00937	0.00	100.953	0.358 0.352	3.744 1.490	3477.8
16.7	97.60	-0.10	0.1	0.444	0.01005	0.31	133.945	0.343	1,910	3477.6 3479.4
								· · · ·	. , ,,,	34/7.4



T2	T5	T 6	Tg	P2	P4		P8	Pg	NOx	HC PPM	CO PPM	RUN
${}_{\rm C}^{\rm 2}$	OC.	T ₆ C	0 €	kPa	kPa 375.2	kPa 183.3	kPa 109.7	kPa 103.9	PPM	1.8	18.1	12
204.4	941.4	791.4 788.7	703.4 701.1	374.9 372.5	372.9	182.3	109.6	103.9		1.7	22.1	33
20 3 . 3 20 1 . 8	937.7 933.7	781.0	697.2	368.2	368.7	180.9	109.4	103.9	78.4	1-9	21.5 20.9	14 35
201.2	928.8	761.6	696.3	365.4	365.9	179.9	109.3 109.1	103.8	78. 1 74. 8	1.9 0.6	22.7	36
198.8	913.9	773.8	690.3	359.0	359.6 354.5	177.7 176.0	109.1	103.7	72.9	0.5	23.2	37
197-6	914.1 914.2	773.0 773.4	688.7 690.4	354.0 349.6	350.3	174.5	108.8	103.7	72.9	0.2	23.9	38
197.2 204.6	941.3	791.1	703.6	174.6	375.5	183.3	109.7	103.9	82.7	1.0	19.9	39 40
206.8	950.7	793.1	769.7	179.0	179.9	184.8	109.9	104.0	84.8 86.4	1.6 1.6	19.6 20.1	41
207.4	954.9	302.3	712.0	301.4	182.1 185.4	196.5	110.0	104.0	91.9	3.4	17.2	4.2
209.3	96+.0	329 . 4	729 . 5	784.5 387.1	389.3	187.1	110.3	134.0	+4.9	3.0	17.3	4.3
211.3	976.3 94 1. 7	7:1.4	704.1	374.5	375., 6	193.7	104.8	103.9	32.2	1.4	21.2	44 45
205.2	945.0	795.4	706.7	376.€	177.7	183	119.8	103.9	33.4 82.8	1.4	20.6 19_3	46
206.9	939.2	787.1	700.5	17).7	380.8 384.9	163.4 181.0	109.9 110.0	104.0	84.2	1.4	20.4	47
209.1	943.3	709.6 778.3	703.3 695.4	383.9 385.4	386.6	181.5	110.1	134.1	82.3	1.4	20.7	48
211.0 211.5	93 3. 9 939 . a	783.7	699.5	396.0	387.3	192.5	110.1	104.1	83.0	1.4	21.5	↓9 50
215.7	941.0	181.4	701.0	387.5	ਾਲਤ ੂਜ	179.	116.2	104.1	32.4	1.4	20.2 19.5	51
217.1	941.8	781.9	704.1	382.7	31.3.0	176.0	110.2	104.0	45.6 84.1	5.0	23.4	54
205.2	942.1	791.4	704.3	178.6 177.f	376.4 375 . 9	183.0 182.5	1.05.3	107.9	72.4	4.7	20.5	55
205.0	423.4	775.6 780.0	693.0 691.7	300.1	273.4	183.5	109.9	103.9	73.7	5.0	21.9	56
206.1 207.6	979.2 936.6	786.2	699.)	383.4	382.9	185.3	110.3	134.0	76.2	5.0	20.6 18.5	57 5ช
208.6	942.2	791.0	702.7	335.7	194.5	105.9	110.2	104.0 134.0	79.6 76.5	5.0 5.1	19.4	59
207.6	940.4	791.4	702.3	343.4	382.3	185.7 185.5	113.9 109.9	103.9	75.6	5.2	19.7	60
205.8	942.2	793.,	704.5 705.1	379.9 377.9	379.0 377.4	135.3	109.4	103.9	76.5	5.1	19.6	6.1
205.6 204.6	942.5 941.1	793.9 741.6	703.9	375.6	375.1	192.4	109.9	104-0	35.3	4.7	22.)	62 63
208.8	947.8	796.4	707.6	395.1	384.7	186.6	110.1	104.0	79-0	4.4	19.4 19.2	64
208.2	944.0	793.2	7.04 . 4	383.7	383.6	186.1	110.1 110.1	104.0 104.0	77_1 76_4	4.4	19.3	65
208.3	944.2	793.2	704.3	383.9 383.2	383.9 383.?	196.2 186.1	110.1	104.0	77.3	4.3	16.5	bń
208-2	942.1 938.2	741.2 788.3	703.1 702.5	376.4	373.6	182.5	139.9	103.9	76.0	3.0	10.6	63
20 3. 6 20 4. 5	942.0	791.9	764.7	377.6	374. 4	192.9	110.0	104.0	78.5	3.1	10.5 11.3	70 71
204.2	902.1	759.2	674.1	378.3	375.8	183.6	199.6 199.6	103.9 103.9	50.4 55.3	2.9 0.6	10.6	72
205.4	904.3	764.4	673.3	380.7 394.8	3 78. 4 382 . 5	184.4 185.9	179.7	133.9	56.6	0.6	10.3	73
207-1	917.8	772.4 781.6	684.3 692.7	398.3	780.1	137.1	109.9	104.0	59.0	J.5	4.1	74
209.0 210.9	928.5 940.9	792.4	701.9	390.€	388.4	187.9	110.0	104.0	62.8	0.5	9.2	75 76
209.3	941.2	744.1	701.0	349.1	386.9	189.7	110.0	104.0	64.2	0.5	8.7 9.9	77
207.1	943.9	798.6	706 - 8	385.2	383.2	189_2 188_8	109.9 109.7	103.9 103.9	62.1	0.1	8.5	78
205.8	944.1	799.9	708.4 702.5	381.9 379.0	380.0 377.1	187.8	109.7	103.9	61.2	0.3	8.8	79
204.7 204.2	935.7 944.8	792.4 802.1	710.0	379.0	377.2	188.3	109.6	103.8	61.3	0.3	8.2	30 81
211.5	938.9	739.8	700.0	391.1	339.4	187-9	110.1	104.0	61.1 63.1	0.3	9.7 7.4	82
212.7	939.4	749.4	697.6	395.7	394.1	189.5 190.2	110.3	104.1 174.1	62.3	0.3	a, j	83
213.1	939.7	789.3	(97 . 1 6 95 . 6	397.? 396.6	795.3 395.1	187.5	110.3	104.1	62.7	0.3	3.8	84
213.2	937.9 939.9	797.4 789.0	696.7	397.6	396.9	190.2	110.4	104.1	61.6	0.3	7.9	85 96
213.7 204.8	395.2	751.8	609.7	377.0	375.7	182.8	105.6	101.9	49.6 79.0	0.3	10.4 9.5	87
204.3	934.3	794.7	609.1	375.3	374.5	182.0 183.1	169.9 109.7	103.9 103.9	85.7	0.9	15.8	90
204.3	942.4	702.5	705.4 (55.1	376.9 377.6	374.4 375.4	133.4	309.1	102.9	44.4	1.3	18.0	91
204.1	876.7 480.3	738.5 741.7	657.4	378.6	376.6	184.2	109.1	103.8	44.3	1.2	18.5	92 93
204.5 206.8	391.4	752.8	666.1	384.4	382.4	136.2	109.3	173.9	43.8 46.6	1.0	18.6 17.3	94
207.4	906 - 1	74.3.4	675.3	384.2	187.3	137_4 188_3	109.4 109.6	103.9 103.9	52.9	0.9	14.9	95
211.4	918.6	774.2	584.5	391.9 393.8	390.1 392.0	194.1	104.8	103.9	57.3	0.9	14.1	90
214.7	941.8 937.0	794.3 789.3	702.7 696.0	430.3	398.5	192.0	110.0	104.0	56.6	0.3	14.2	97
215.2 216.3	938.8	790.1	645.5	404.1	402.4	193.7	110.2	134.1	56.5	1.0	19.5 15.5	99 99
215.9	936.8	788.5	693.1	403.6	402.1	193.4	110.1	104.1	56.1 56.9	1.1	15.6	100
216.6	937.6	798.7	694.4 407.0	404.6	405.8 405.8	194.0	110.2	104.1	57.9	0.9	14.9	101
217.8	943.8	793.9 796.5	697.0 703.2	407.3 379.1	377.7	190.6	109.2	103.8	54.0	0.9	14.2	102
204.4 205.1	934.8 93 7. 1	798.3	704.1	381.3	380.0	191.2	109.3	103-9	53.5	0.8	15.8 12.9	103 104
208.0	942.7	301.3	705.8	387.4	306.2	192.4	109.5	103.8 103.9	55.8 55.3	0.9 0.9	15.6	105
210.3	940.4	797.1	703.4	391.4	390.1	192.2 191.2	109.7 109.7	103.9	55.6	0.9	12.8	106
210.7	936.8	793.1	701.6	390.2 392.1	389.0 39 1. 1	191.7	109.3	103.9	55.7	0.9	14.7	107
211.4	941.2 942.9	797.0 793.0	706.0	375.2	373_8	182.7	109.5	103.6				164 165
209.2	961.3	808.1	717.7	384.2	382.7	186.0	117.0	103.7 103.9				160
212.0	973.0	821.4	727.9	393.9	392 .7 395.3	189.7 190.5	110.5 110.6	103.9				167
214.2	987.5	429.0	734.2	396.6	177.5	170 .)						

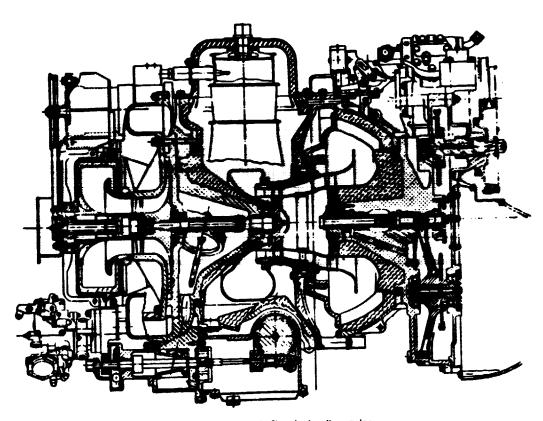


Figure 1. - Schematic Chrysler teseline engine.

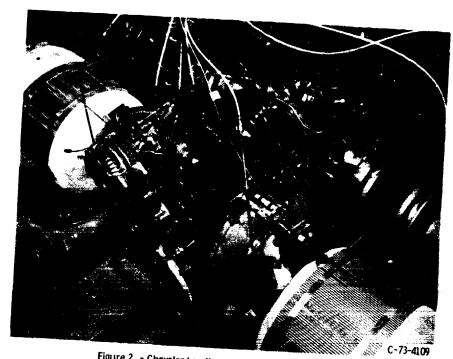


Figure 2. - Chrysler baseline engine installed in test cell.



Figure 3. - Picture of water nozzle location on engine.

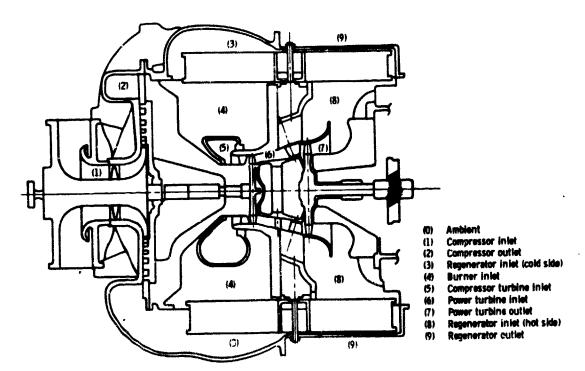


Figure 4. - Chrysler baseline engine station notation.

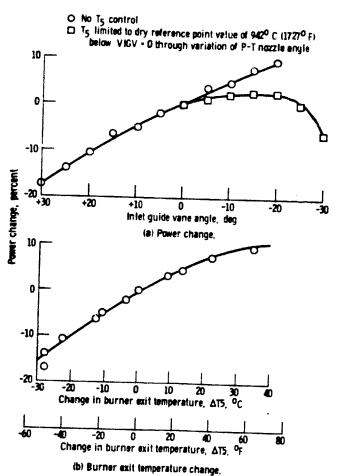


Figure 5. - Effect of varying VIGV angle at 95 percent $\,N_{GG}$ - no water injection.

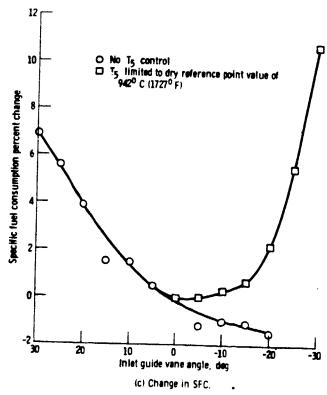


Figure 5. - Concluded.

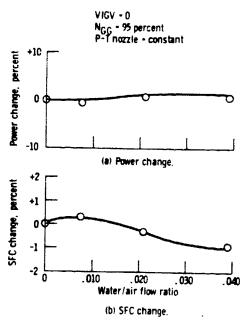
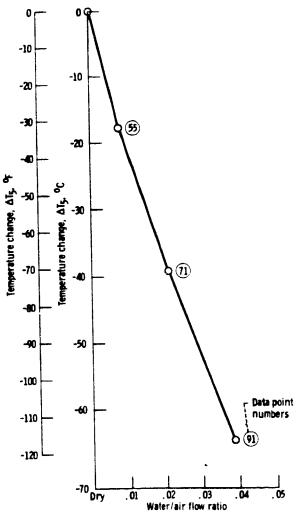


Figure 6. - Effect of water injection.

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(c) Burner exit temperature change. Figure 6. - Concluded.

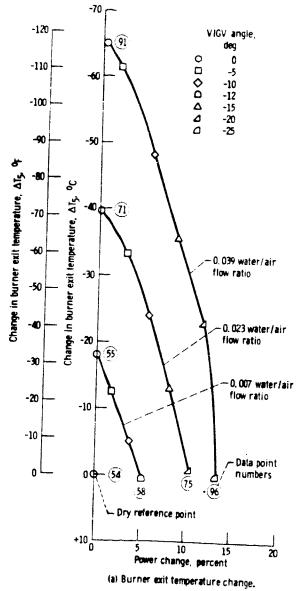


Figure 7. - Effect of varying both water injection and VIGV angle,

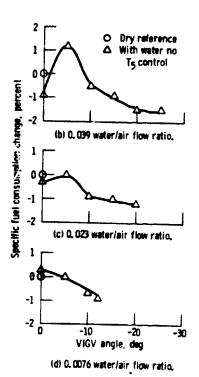


Figure 7. - Concluded.

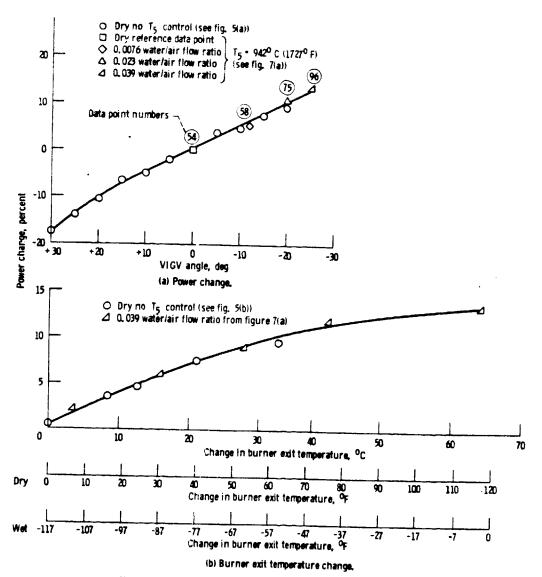


Figure 8. - Comparison of ViGV angle effect with and without water injection.

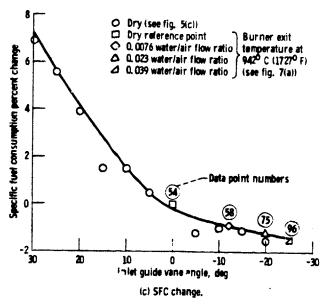


Figure & - Concluded.

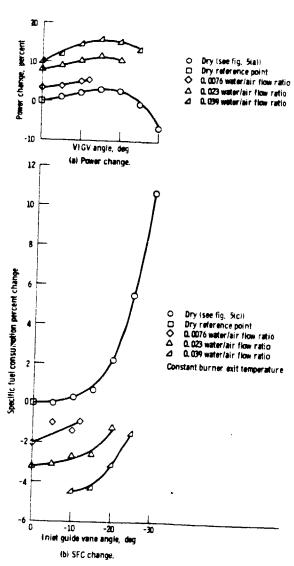


Figure 9, - Effect of variation of water injection and Vt GV angle at 95 percent gas generator speed with burner exit temperature held constant by varying power turbine nozzle angle.

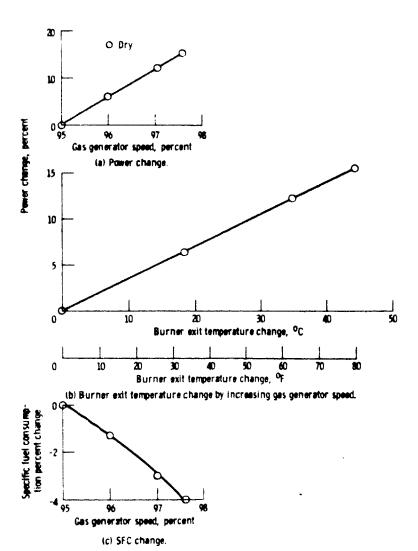


Figure 10. - Effect of varying gas generator speed (without water injection, 0 ViGV angle, constant power turbine nozzle angle).

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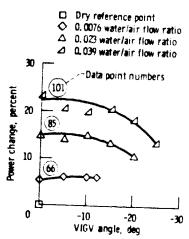


Figure 11. - Effect of variation of water injection and VIGV angle with burner exit temperature held constant by varying gas generator speed.

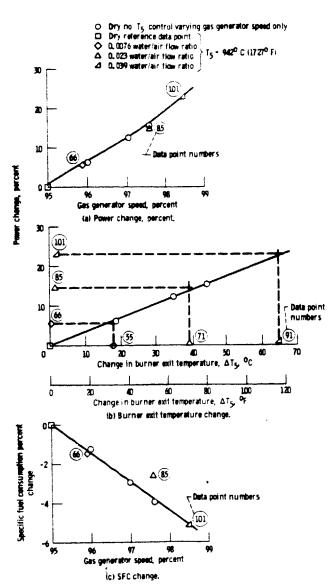
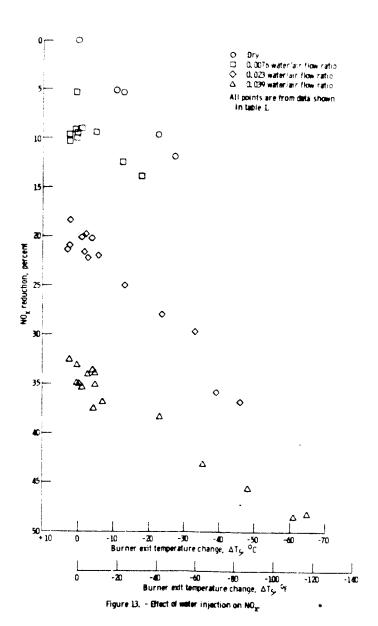


Figure 12. - Comparison of gas generator speed effect with and without water.



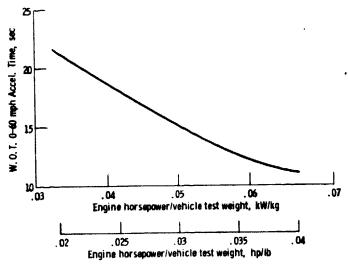


Figure 14. - Typical Auto, Accel. Time vs hp/wt Ratio: 3 Speed Automatic Transmission.

OF PRIOR CHALPS

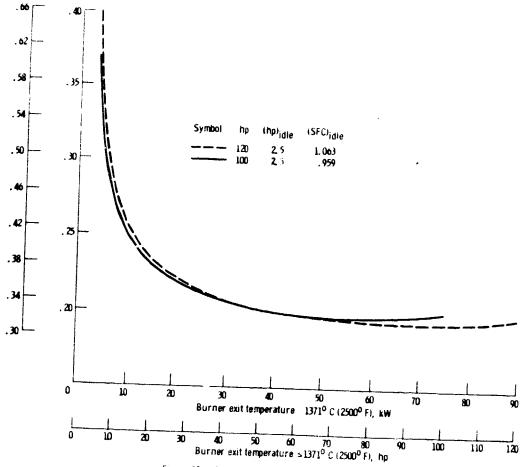


Figure 15. - Effect of design power on SFC's,

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